

Control Systems I

Recitation 08

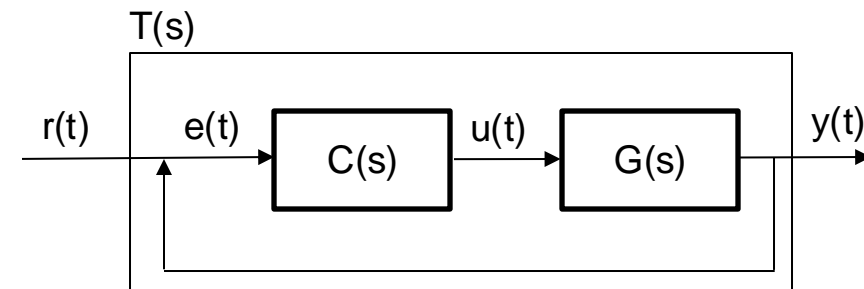
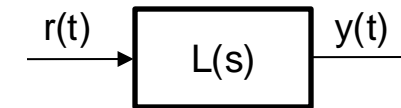
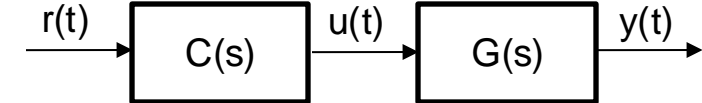
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Controllers and Feedback

Introduction

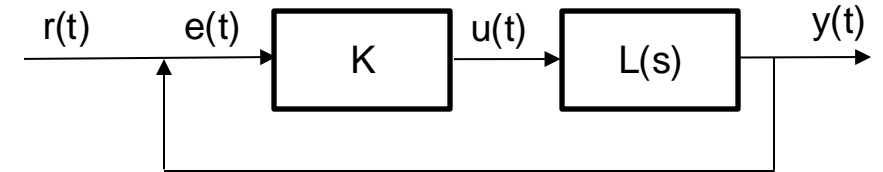
- Until now we only looked at the system: $u(t) \rightarrow y(t)$
 - Stability and Analysis until now only for this
- We can introduce a Controller (no feedback yet):
 - **Open Loop System:** $r(t) \rightarrow y(t)$
 - $L(s) = C(s)G(s)$
- We can Introduce Feedback:
 - **Closed Loop System:** $r(t) \rightarrow y(t)$
 - $T(s) = \frac{L(s)}{1+L(s)} = \frac{C(s)G(s)}{1+C(s)G(s)}$
 - Changes the system behaviour dynamically
 - Unstable \rightarrow stable
 - Stable \rightarrow “More” stable (quicker or less oscillation)
 - Stable \rightarrow unstable



Root Locus

What?

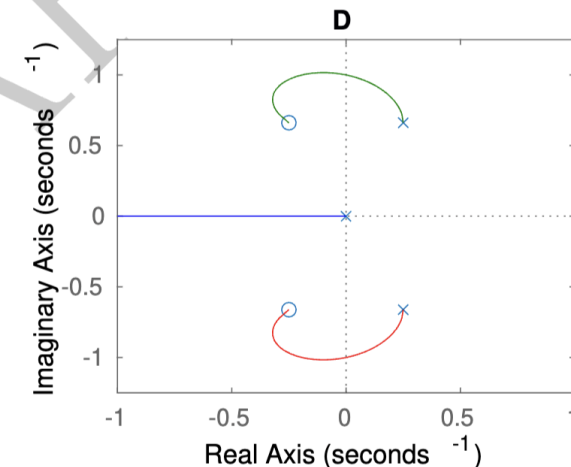
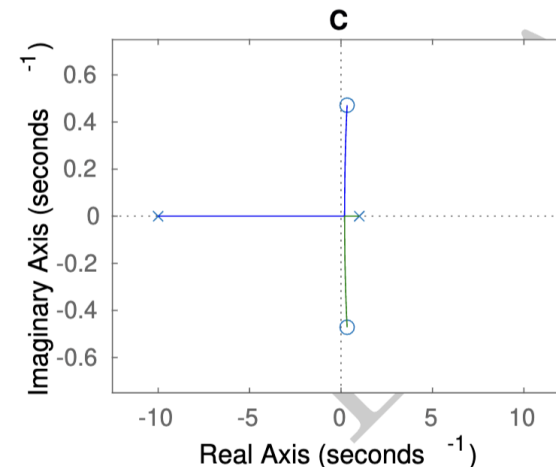
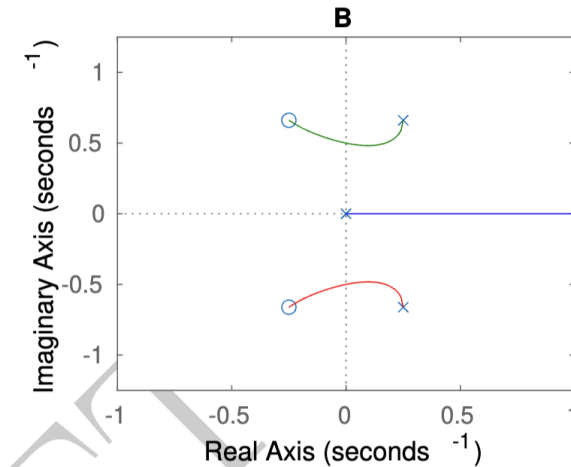
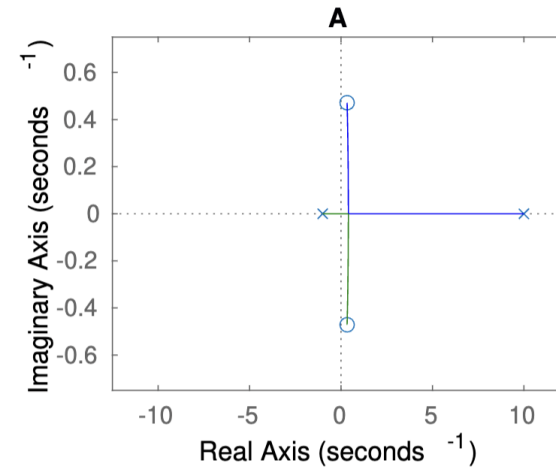
- Add proportional controller to the open loop system:
 - $kL(s) = k \frac{N(s)}{D(s)} \rightarrow T(s) = \frac{kL(s)}{1+kL(s)} = \frac{kN(s)}{D(s)+kN(s)}$
 - $kL(s) = k \frac{(s-z_1)(s-z_2)\dots(s-z_m)}{(s-p_1)(s-p_2)\dots(s-p_n)}$
- Root Locus = Graphical analysis of the closed loop system for different values of k or any other parameter
 - Plot the position of the zeros and poles for all possible k
- Using only the **open loop system** to analyse the **closed loop system**!
 - We only need to know $L(s)$!!!
 - Get quick (qualitative) info about the system response
- Different extremes:
 - $k = 0$: Poles of $T(s) \rightarrow$ Poles of $L(s)$
 - $k \rightarrow \infty$: Poles of $T(s) \rightarrow$ Zeros of $L(s)$
 - This also explains why we should avoid non-minimum phase zeros
 - Since degree of $N(s)$ is smaller than $D(s)$ the "excess" poles go to ∞



Root Locus

Extracting Information

- Get Open Loop Transfer Function:
 - Zeros are circles
 - Poles are crosses
 - $$L(s) = \frac{(s-z_1)(s-z_2)\dots(s-z_m)}{(s-p_1)(s-p_2)\dots(s-p_n)}$$
- Stability:
 - Check for poles with positive real part
 - Zeros with positive real part mean that the system becomes unstable if k to large
- Damping and Oscillation (poles and zeros):
 - Imaginary parts -> Oscillation
 - Negative real part -> Damping

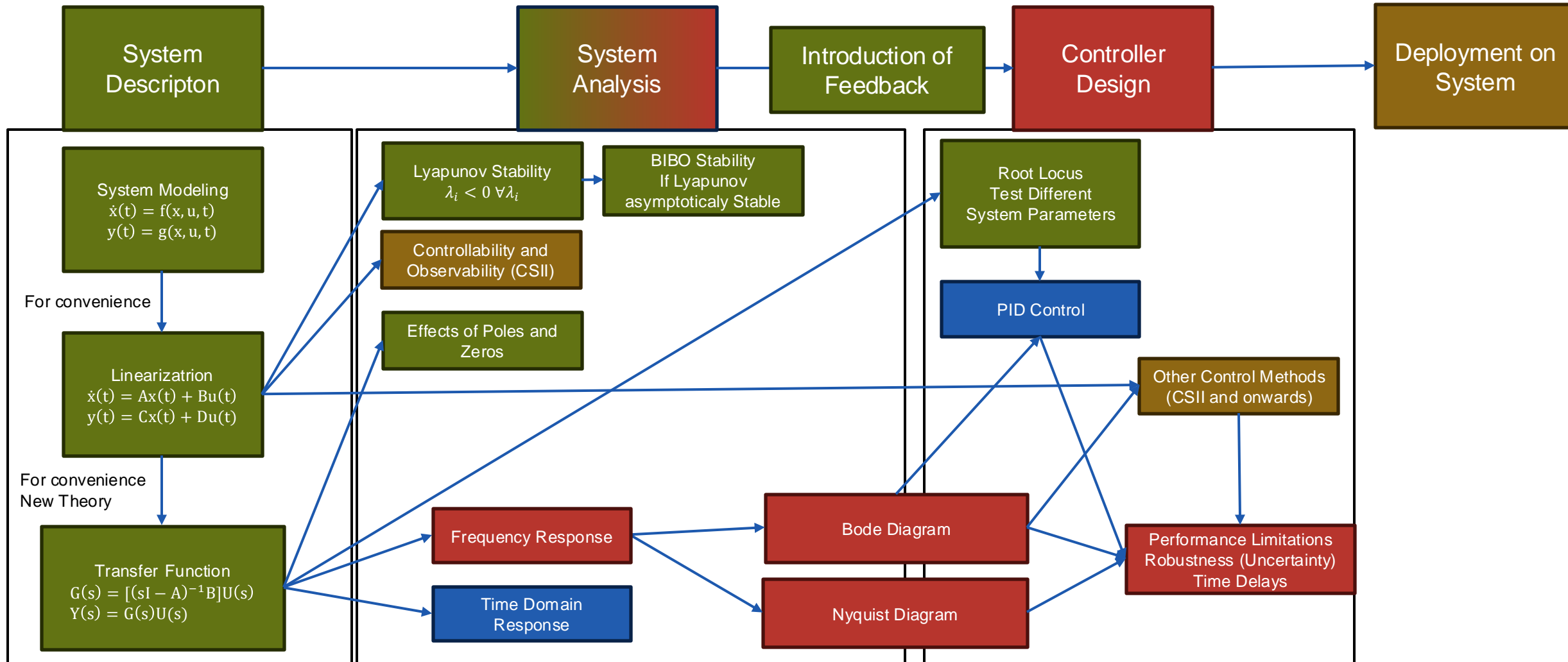


Outline

- System Analysis
 - What is there to check?
 - Different Transfer Functions
 - Sidenote on Total System Behaviour
- Time Response
 - Steady State Error
 - Example
 - Step Response: 1 order system
 - Step Response: 2 order system
- PID Controller
 - Proportional Part
 - Integral Part
 - Derivative Part
 - PID Tuning

Conceptual Recap

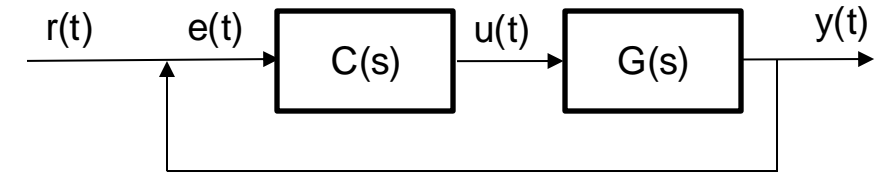
Classical Control Approach



System Analysis

What is there to check?

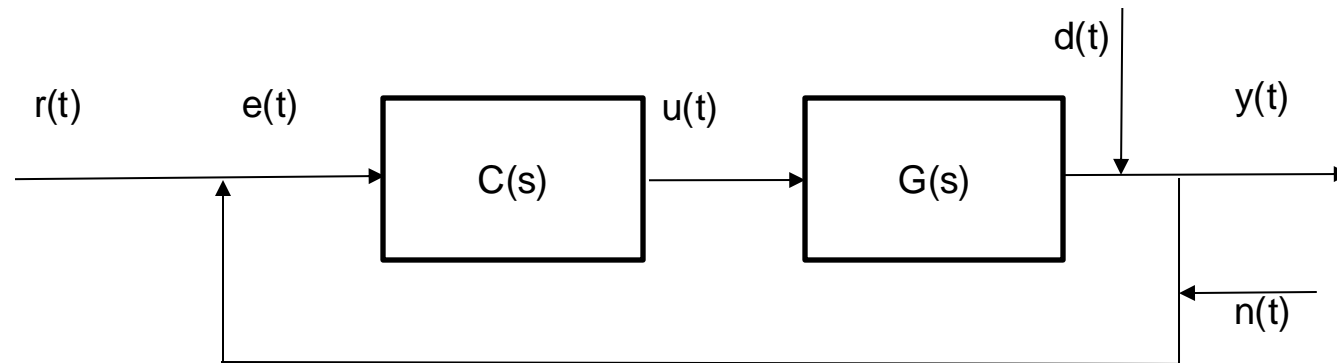
- Qualitative Analysis: (already seen)
 - Stability Analysis: Lyapunov, BIBO, Root Locus
- Quantitative Analysis: (today)
 - Steady State behaviour
 - Do we converge to the desired value?
 - Speed of Convergence
 - How fast do we converge?
 - Overshoot
 - Do we have oscillation and if so, how big are these?
- How about robustness? (next week)
 - Modelling errors
 - If the nominal system is stable is the real system also stable?
 - How big of an error can the system compensate for?
 - Noise/Disturbance Rejection
 - If we have disturbances/noise how good can the closed loop system handle these?



System Analysis

Different Transfer Functions

- We saw the Closed Loop System:
 - **Complementary Sensitivity:** $r(t) \rightarrow y(t)$ or $n(t) \rightarrow y(t)$
 - $T(s) = \frac{Y(s)}{N(s)} = \frac{Y(s)}{R(s)} = \frac{L(s)}{1+L(s)} = \frac{C(s)G(s)}{1+C(s)G(s)}$
- There are more:
 - **Sensitivity:** $r(t) \rightarrow e(t)$ or $d(t) \rightarrow y(t)$
 - $S(s) = \frac{E(s)}{R(s)} = \frac{Y(s)}{D(s)} = \frac{1}{1+L(s)} = \frac{1}{1+C(s)G(s)}$
- We also see that $S(s) + T(s) = 1$
 - We can either track the reference perfect or reject noise perfect



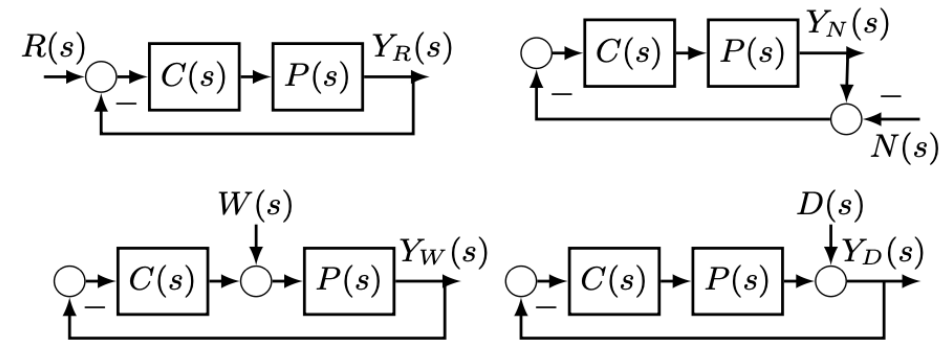
System Analysis

Sidenote on Total System Behaviour

- We can do this for all possible inputs to the system:
- **We get:**

$$Y_R(s) = \frac{P(s)C(s)}{1 + P(s)C(s)} \cdot R(s), \quad Y_N(s) = \frac{P(s)C(s)}{1 + P(s)C(s)} \cdot N(s)$$

$$Y_W(s) = \frac{P(s)}{1 + P(s)C(s)} \cdot W(s), \quad Y_D(s) = \frac{1}{1 + P(s)C(s)} \cdot D(s)$$



- Linearity allows us to: $Y(s) = Y_R(s) + Y_N(s) + Y_W(s) + Y_D(s)$
 - We thus get: $T(s) = \frac{L(s)}{1+L(s)}, S(s) = \frac{1}{1+L(s)} \rightarrow$

$$Y(s) = S(s) \cdot [D(s) + P(s) \cdot W(s)] + T(s) \cdot [R(s) + N(s)]$$
- Learnings:
 - System is stable iff $\frac{1}{1+L(s)}$ is stable

Time Response

Steady State Error

- Error: $e(t) = r(t) - y(t)$ should ideally go to zero $\lim_{t \rightarrow \infty} e(t) = 0$
- Sensitivity describes the relation between $r(t) \rightarrow e(t)$: $E(s) = S(s)R(s) = \frac{1}{1+L(s)} R(s)$
- For a step input $R(s) = \frac{1}{s} \rightarrow E(s) = \frac{1}{s} \frac{1}{1+L(s)}$
 - Steady state behaviour:
 - $e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{1}{1+L(s)} = \frac{1}{1+L(0)}$
 - How can we make sure we have no steady state error?
 - $L(0) \rightarrow \infty \rightarrow$ need a pole at $s = 0$ (integrator)
 - If the closed loop system is stable and we have an integrator we have 0 steady state error
 - How do we get $L(0)$?
 - $L(s) = \frac{k_{bd}}{s^q} \frac{\left(\frac{s}{-z_1}+1\right)\left(\frac{s}{-z_2}+1\right)\dots\left(\frac{s}{-z_1}+1\right)}{\left(\frac{s}{-p_1}+1\right)\left(\frac{s}{-p_2}+1\right)\dots\left(\frac{s}{-p_{n-q}}+1\right)} \xrightarrow{s \rightarrow 0} L(0) \rightarrow \frac{k_{bd}}{s^q}$
 - q is called type $L(s)$ for $q = 0$ we get $L(0) = k_{bd}$ which is called the DC-gain
 - $q > 0$: $L(0) \rightarrow \infty$

Time Response

Steady State Error

- Steady State error for a step input $R(s) = \frac{1}{s} \rightarrow E(s) = \frac{1}{s} \frac{1}{1+L(s)}$
 - $e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{1}{1+L(s)} = \frac{1}{1+L(0)}$
- What about ramps $r(t) = \frac{1}{q!} t^q \rightarrow R(s) = \frac{1}{s^{q+1}}$?
 - $e_{ss} = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} sE(s) = \lim_{s \rightarrow 0} \frac{1}{s^q} \frac{1}{1+L(s)}$
 - To have 0 ss-error we need at least one integrator more in our system

e_{ss}	$q = 0$	$q = 1$	$q = 2$
Type 0	$\frac{1}{1 + k_{Bode}}$	∞	∞
Type 1	0	$\frac{1}{k_{Bode}}$	∞
Type 2	0	0	$\frac{1}{k_{Bode}}$

- The same derivations go for disturbances!

Time Response

Steady State Error - Example

- Given the system: $G(s) = \frac{s+3}{s^3+5s+3}$ and a Controller $C(s) = 3 + 6s$
- What is the steady state error for a step input a linear ramp?
 -
 -
 - Step input:
 - Impulse input:
 -
 -
 - Propose a controller that we get a zero ss error:
 -

Time Response

Step Response: 2 order system

- First order system:

- $\dot{x}(t) = -\frac{1}{\tau}x(t) + \frac{k}{\tau}u(t), y(t) = x(t)$

- $G(s) = \frac{k}{\tau s + 1}$

- Time response to a step input:

- $y(t) = x_0 e^{-\frac{t}{\tau}} + k(1 - e^{-\frac{t}{\tau}})$

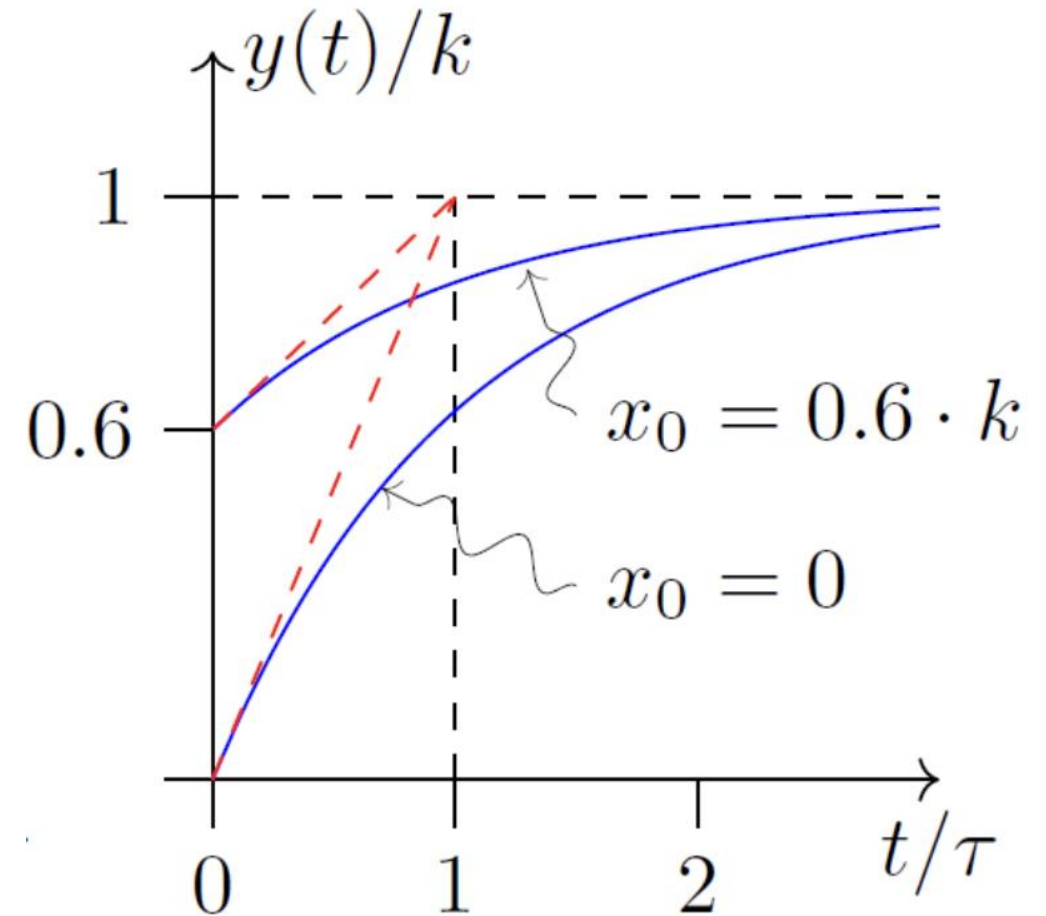
- What we can see:

- x_0 determines the start value
- DC-gain/Steady State: $y_{ss} = k$
- Evaluating the derivative at $t = 0$:
 - The tangent crosses $y = k$ at $t = \tau$

- Large τ slow convergence

- Settling Time: Time it takes to be at d% of y_{ss} (beware this is the opposite as in the lecture)

- $T_d = \tau \ln\left(\frac{100}{1-d}\right) \rightarrow \tau = \frac{T_d}{\ln\left(\frac{100}{1-d}\right)}$



Time Response

Step Response: 2 order system

- Second order system:

- $\dot{\mathbf{x}}(t) = \begin{bmatrix} 0 & 1 \\ -\omega_n^2 & -2\zeta\omega_n \end{bmatrix} \mathbf{x}(t) + \begin{bmatrix} 0 \\ \omega_n^2 \end{bmatrix} u(t), \quad y(t) = \mathbf{x}(t)$

- $G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$

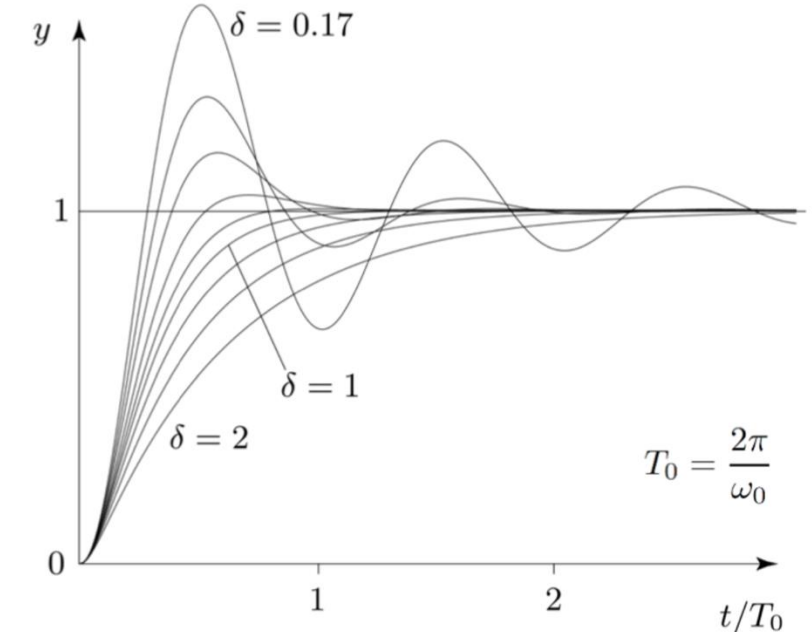
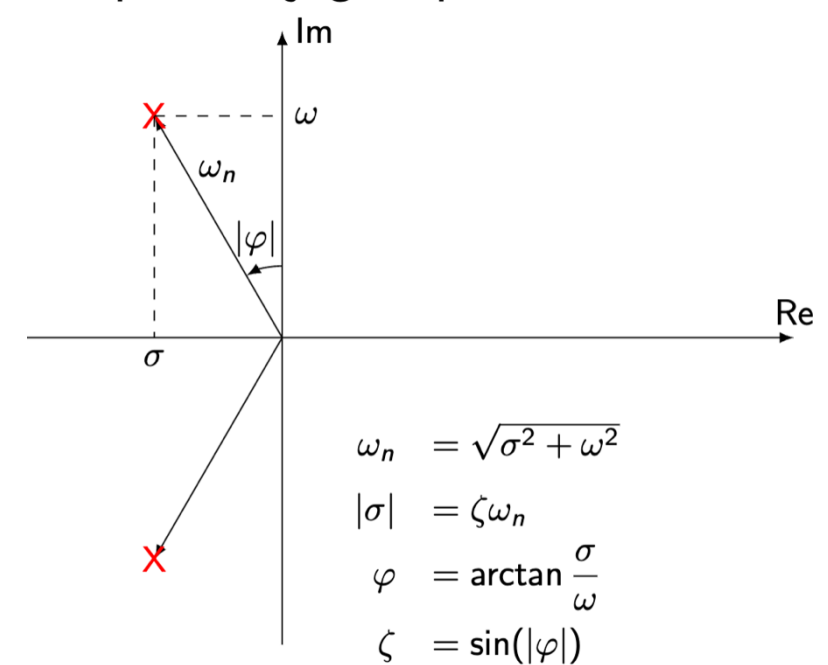
- Time response to a step input (depends on ζ): (in Mechanics III)

- Underdamped ($\zeta < 1$):

- $y(t) = 1 - \frac{1}{\cos(\varphi)} e^{\sigma t} \cos(\omega t + \varphi)$

- $\varphi = \arctan\left(\frac{\zeta}{\omega}\right), \quad \sigma = \zeta\omega_n, \quad \omega = \sqrt{\omega_n^2 - (\zeta\omega_n)^2} = \omega_n \sqrt{1 - \zeta^2}$

- Overdamped ($\zeta > 1$): We get cosh() and sinh() terms (not relevant for you)



Time Response

Step Response: 2 order system

- Time response to a step input underdamped ($\zeta < 1$):

- $$y(t) = 1 - \frac{1}{\cos(\varphi)} e^{\sigma t} \cos(\omega t + \varphi)$$

- $$\varphi = \arctan\left(\frac{\zeta}{\omega}\right), \quad \sigma = \zeta\omega_n, \quad \omega = \sqrt{\omega_n^2 - (\zeta\omega_n)^2} = \omega_n\sqrt{1 - \zeta^2}$$

- What can we specify?

- Settling time: (on exponential envelope)

- $$T_d = \frac{1}{\sigma} \ln\left(\frac{100}{1-d}\right) \rightarrow \sigma = \frac{\ln\left(\frac{100}{1-d}\right)}{T_d}$$

- Time to peak:

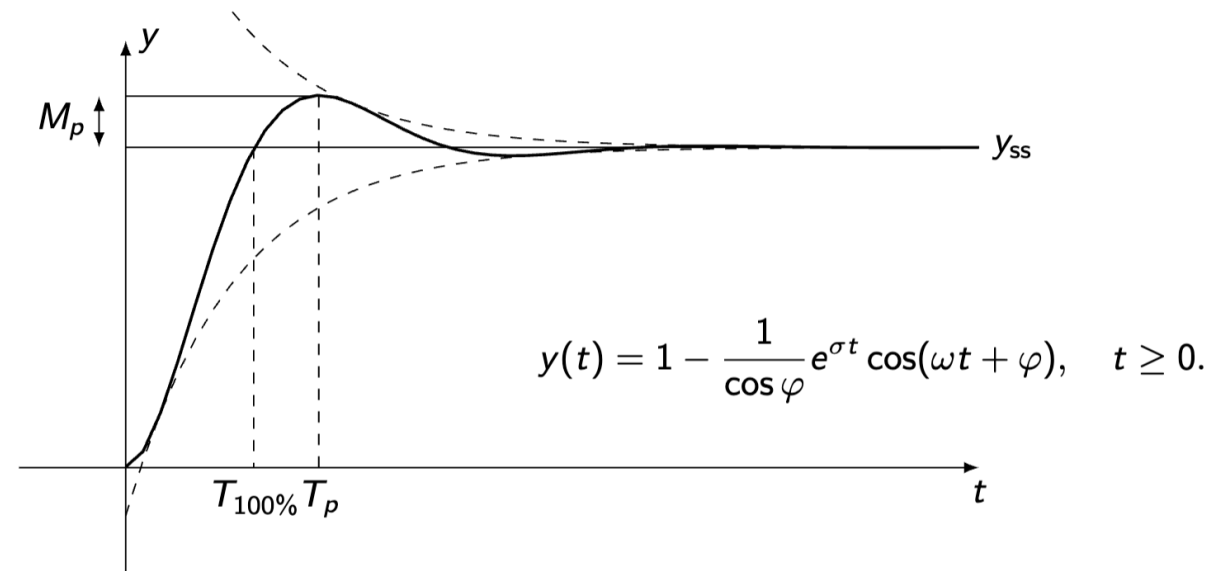
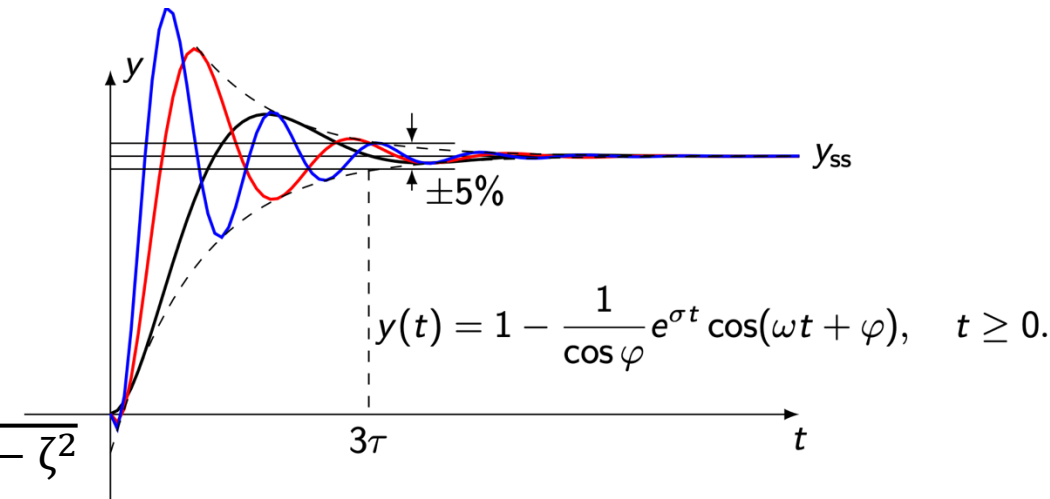
- $$T_p = \frac{\pi}{\omega}$$

- Overshoot: (low damping = high overshoot)

- $$M_p = e^{\frac{\sigma\pi}{\omega}} \rightarrow \zeta^2 = \frac{(\ln(M_p))^2}{\pi^2 + (\ln(M_p))^2}$$

- Rise Time:

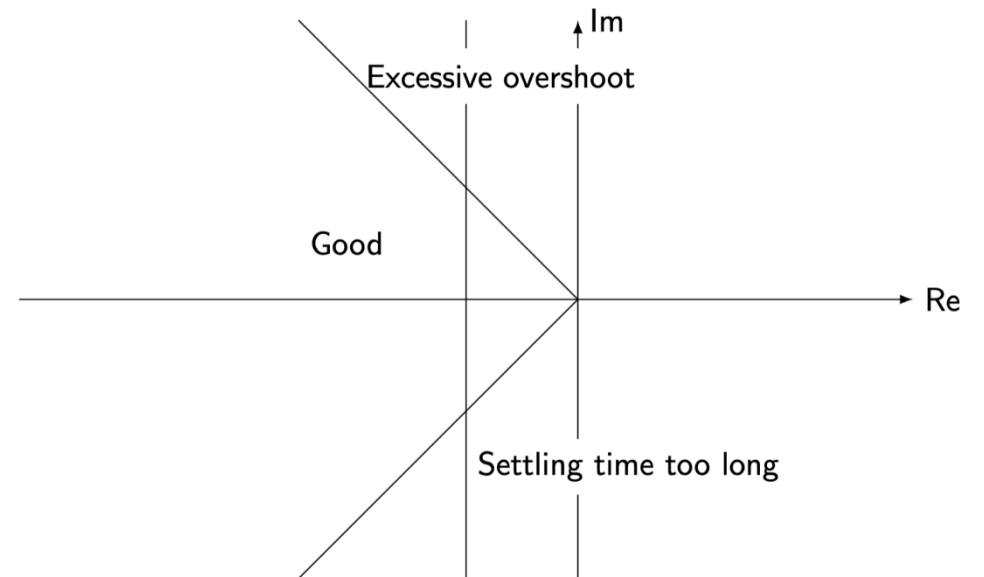
- $$T_{100} = \frac{\frac{\pi}{2} - \varphi}{\omega} \approx \frac{\pi}{2\omega_n}$$



Time Response

Step Response: Higher Order Systems

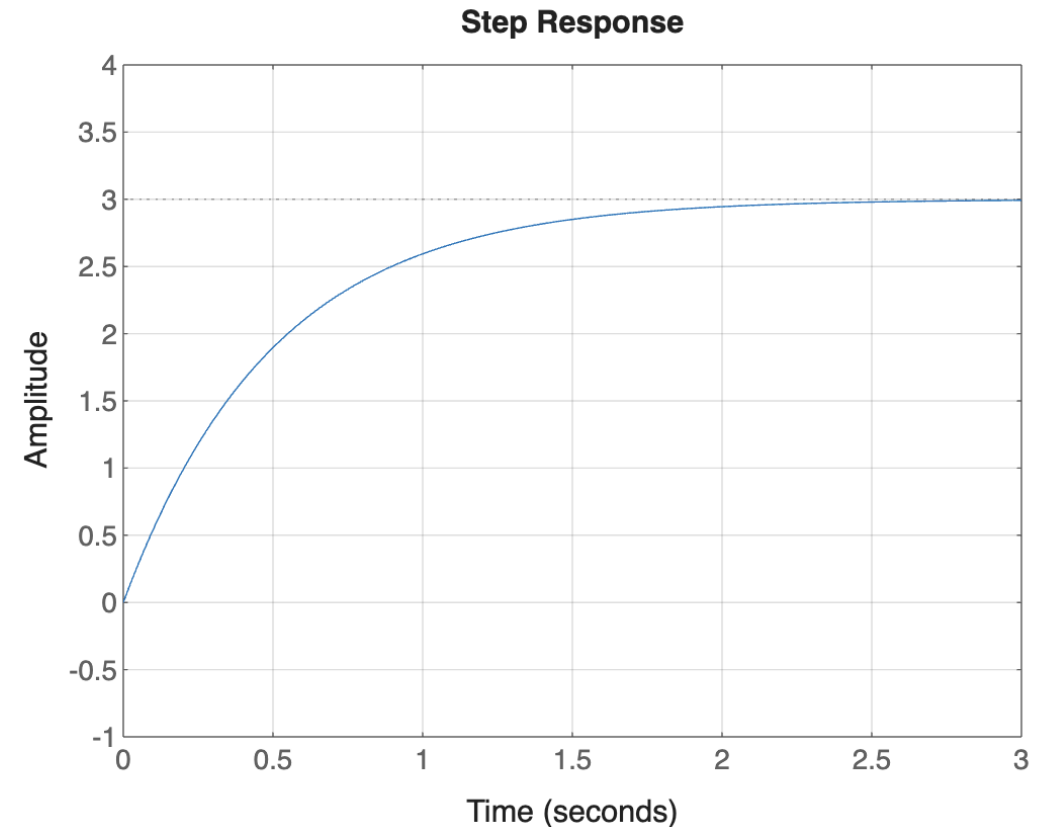
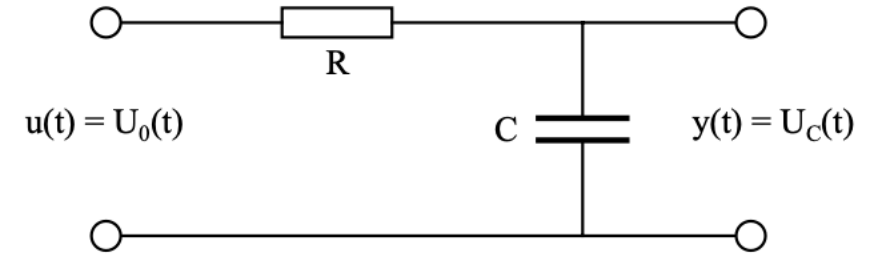
- What about higher order systems?
 - Approximate them with a 1 or 2 order system
 - Apply the specifications to the approximation
- What poles to choose?
 - No zeros: Chose the poles with the slowest decay time
 - With zeros: Highest residuals
- In General:
 - We want high real part (fast decay)
 - We want low imaginary part (smaller overshoot)
 - We can see this in the Imaginary Plane



Time Response

Step Response: Example

- We have a electrical system that can be described by:
 - $u(t) = RC\dot{y}(t) + y(t), R = 1$
- We got the plot of a step input voltage:
 - $u(t) = a \cdot h(t)$
- What is the value of C?
 -
 -
 -
- What are the value of a?
 -
- If we choose $R = 2$ what does C need to be for an identical system behaviour?
 -



PID Control

Proportional Part

- General Formulation:

- $u(t) = k_p e(t) + k_d \dot{e}(t) + k_i \int e(t) dt$
- $U(s) = \left(k_p + k_d s + \frac{k_i}{s} \right) E(s) = C(s) E(s)$

- We get: $T(s): r(t) \rightarrow y(t)$ or $n(t) \rightarrow y(t)$, $S(s): r(t) \rightarrow e(t)$ or $d(t) \rightarrow y(t)$

- $T(s) = \frac{\left(k_p + k_d s + \frac{k_i}{s} \right) G(s)}{1 + \left(k_p + k_d s + \frac{k_i}{s} \right) G(s)}, \quad S(s) = \frac{1}{1 + \left(k_p + k_d s + \frac{k_i}{s} \right) G(s)}$

- Proportional Part:

- $T(s) = \frac{(k_p)G(s)}{1 + (k_p)G(s)}, \quad S(s) = \frac{1}{1 + (k_p)G(s)}$
- High k_p : $T(s) \rightarrow 1, S(s) \rightarrow 0$
 - Faster response
 - Lower steady state error
 - Higher sensitivity to noise

PID Control

Integral Part

- General Formulation:
 - $u(t) = k_p e(t) + k_d \dot{e}(t) + k_i \int e(t) dt$
 - $U(s) = \left(k_p + k_d s + \frac{k_i}{s} \right) E(s) = C(s) E(s)$
- Proportional and Integrator Part:
 - As seen before integrator part reduces steady state error to 0
 - High k_i :
 - System starts to oscillate
 - Integrator fills up and needs to be emptied on the other side
 - Still sensitive to noise

PID Control

Derivative Part

- General Formulation:
 - $u(t) = k_p e(t) + k_d \dot{e}(t) + k_i \int e(t) dt$
 - $U(s) = \left(k_p + k_d s + \frac{k_i}{s} \right) E(s) = C(s) E(s)$
- Proportional, Derivative and Integrator Part:
 - High k_d :
 - Steady state error not affected
 - Less oscillation (more damping)
 - Slows down the system
 - Sensitivity to noise increases
 - High frequency noise derivatives are large

PID Control

Summary

- Proportional Part:
 - Faster response
 - Lower steady state error
 - Higher sensitivity to noise
 - Reduces stability margin -> can make system unstable
- Integrator Part:
 - Eliminates steady state error (step input)
 - Introduces oscillation
 - Reduces stability margin -> can make system unstable
- Derivative Part
 - Reduces Overshooting, Increases Damping
 - Improves stability margins
 - Very sensitive to noise
 - Not physically realizable (use approximation)

PID Control

PID Design/PID Tuning

- Freestyle:
 - Start with k_p add a bit of k_d to dampen and add k_i to remove ss-error, see what works
- Root Locus in the lecture
 - Recursively changing the values of k_p, k_d, k_i
- Ziegler-Nichols: systematic approach
 - Increase k_p until the system becomes marginally stable (start oscillating without decay)
 - Get k_p^* and $T^* = \frac{\omega^*}{2\pi}$
- Aström-Hägglund: systematic approach
 - Get k_p^* and T^* like Ziegler-Nichols
 - Get $|P(0)|$ using measurements of a step response
- Optimal Design and Stability not guaranteed
 - Real world testing often needed

Regler	k_p	T_i	T_d
P	$0.5 \cdot k_p^*$	$\infty \cdot T^*$	$0 \cdot T^*$
PI	$0.45 \cdot k_p^*$	$0.85 \cdot T^*$	$0 \cdot T^*$
PD	$0.55 \cdot k_p^*$	$\infty \cdot T^*$	$0.15 \cdot T^*$
PID	$0.60 \cdot k_p^*$	$0.50 \cdot T^*$	$0.125 \cdot T^*$

$$\kappa = \frac{1}{|P(0)| \cdot k_p^*}, \quad x = \alpha_{0,x} \cdot e^{\alpha_{1,x} \cdot \kappa + \alpha_{2,x} \cdot \kappa^2}$$

x	$\mu_{\min} = 0.7$			$\mu_{\min} = 0.5$		
	$\alpha_{0,x}$	$\alpha_{1,x}$	$\alpha_{2,x}$	$\alpha_{0,x}$	$\alpha_{1,x}$	$\alpha_{2,x}$
$\frac{k_p}{k_p^*}$	0.053	2.90	-2.6	0.13	1.9	-1.30
$\frac{T_i}{T^*}$	0.900	-4.40	2.7	0.90	-4.4	2.70

x	$\mu_{\min} = 0.7$			$\mu_{\min} = 0.5$		
	$\alpha_{0,x}$	$\alpha_{1,x}$	$\alpha_{2,x}$	$\alpha_{0,x}$	$\alpha_{1,x}$	$\alpha_{2,x}$
$\frac{k_p}{k_p^*}$	0.33	-0.31	-1.00	0.72	-1.60	1.20
$\frac{T_i}{T^*}$	0.76	-1.60	-0.36	0.59	-1.30	0.38
$\frac{T_d}{T^*}$	0.17	-0.46	-2.10	0.15	-1.40	0.56

Exercise 07

What to do?

- 1:
 - A) do
 - B) look at solution and think about the answer
 - C) look at solution and think about the answer
 - D) optional
 - E) play around a bit
- 2:
 - A) optional
 - B) do, need solution of A)
 - C) do, cumbersome but good practice
 - D) do
- 3:
 - A) do
 - B) do
 - C) optional